



TOBB EKONOMİ VE TEKNOLOJİ ÜNİVERSİTESİ
MAK 413 MECHANICS OF COMPOSITE MATERIALS



SPRING 2018

Due Date: 05.02.2018- Monday* (08:30)

HOMEWORK 1

1. Kaw 2nd ed. Prob. 2.2

2.2 The engineering constants for an orthotropic material are found to be

$$E_1 = 4 \text{ Msi}, E_2 = 3 \text{ Msi}, E_3 = 3.1 \text{ Msi},$$

$$\nu_{12} = 0.2, \nu_{23} = 0.4, \nu_{31} = 0.6,$$

$$G_{12} = 6 \text{ Msi}, G_{23} = 7 \text{ Msi}, G_{31} = 2 \text{ Msi}$$

Find the stiffness matrix [C] and the compliance matrix [S] for the preceding orthotropic material.

2. Kaw 2nd ed. Prob. 2.3

2.3 Consider an orthotropic material with the stiffness matrix given by

$$[C] = \begin{bmatrix} -0.67308 & -1.8269 & -1.0577 & 0 & 0 & 0 \\ -1.8269 & -0.67308 & -1.4423 & 0 & 0 & 0 \\ -1.0577 & -1.4423 & 0.48077 & 0 & 0 & 0 \\ 0 & 0 & 0 & 4 & 0 & 0 \\ 0 & 0 & 0 & 0 & 2 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1.5 \end{bmatrix} \text{ GPa}$$

Find:

1. The stresses in the principal directions of symmetry if the strains in the principal directions of symmetry at a point in the material are $\epsilon_1 = 1 \mu\text{m}/\text{m}$, $\epsilon_2 = 3 \mu\text{m}/\text{m}$, $\epsilon_3 = 2 \mu\text{m}/\text{m}$; $\gamma_{23} = 0$, $\gamma_{31} = 5 \mu\text{m}/\text{m}$, $\gamma_{12} = 6 \mu\text{m}/\text{m}$
2. The compliance matrix [S]
3. The engineering constants $E_1, E_2, E_3, \nu_{12}, \nu_{23}, \nu_{31}, G_{12}, G_{23}, G_{31}$
4. The strain energy per unit volume at the point where strains are given in part (1.)

3. Kaw 2nd ed. Prob. 2.10

2.10 Find the reduced stiffness $[Q]$ and the compliance $[S]$ matrices for a unidirectional lamina of boron/epoxy. Use the properties of a unidirectional boron/epoxy lamina from [Table 2.1](#).

4. Kaw 2nd ed. Prob. 2.11

2.11 Find the strains in the 1–2 coordinate system (local axes) in a unidirectional boron/epoxy lamina, if the stresses in the 1–2 coordinate system applied to are $\sigma_1 = 4$ MPa, $\sigma_2 = 2$ MPa, and $\tau_{12} = -3$ MPa. Use the properties of a unidirectional boron/epoxy lamina from [Table 2.1](#).

5. Kaw 2nd ed. Prob. 2.14

2.14 Consider a unidirectional continuous fiber composite. Start from $[\sigma] = [Q][\epsilon]$ and follow the procedure in [Section 2.4.3](#) to get

$$E_1 = Q_{11} - \frac{Q_{12}^2}{Q_{22}} \quad \nu_{12} = \frac{Q_{12}}{Q_{22}}$$

$$E_2 = Q_{22} - \frac{Q_{12}^2}{Q_{11}} \quad \nu_{21} = \frac{Q_{12}}{Q_{11}} \quad G_{12} = Q_{66}$$

Due date is **Monday 5th of February, 2018**. For each delayed day **15 points** will be reduced.